

## High Chamber Pressure, Light Weight Thrusters, Phase I

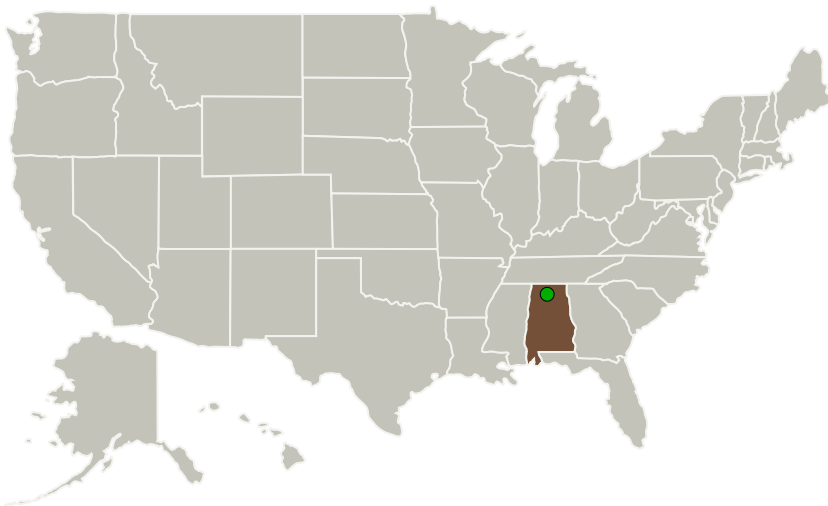
Completed Technology Project (2010 - 2010)



## Project Introduction

The performance liquid propellant engines can be significantly improved by increasing both combustion temperature and pressure and reducing engine weight. State of the art Bi-propellant engines are metallic; the highest temperature combustion chambers are made of iridium/rhenium metals. Utilizing carbon-carbon composites for thrust chamber structures can reduce component weight by as much as 90%. Unfortunately carbon composites suffer from unacceptably high oxidation rates. Lining carbon-carbon chambers with a thin layer of iridium or iridium/rhenium is an innovative way to provide the oxidation barrier necessary to enable the use of carbon-carbon composites. This proposal evaluates using metal-lined carbon-carbon to produce high pressure, light weight thrust chambers.

## Primary U.S. Work Locations and Key Partners



| Organizations Performing Work         | Role                    | Type        | Location            |
|---------------------------------------|-------------------------|-------------|---------------------|
| ● Marshall Space Flight Center (MSFC) | Supporting Organization | NASA Center | Huntsville, Alabama |

## Primary U.S. Work Locations

Alabama



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## Table of Contents

|  |   |
|--|---|
| Project Introduction                         | 1 |
| Primary U.S. Work Locations and Key Partners | 1 |
| Organizational Responsibility                | 1 |
| Project Transitions                          | 2 |
| Project Management                           | 2 |
| Technology Maturity (TRL)                    | 2 |
| Technology Areas                             | 2 |
| Target Destinations                          | 2 |

## Organizational Responsibility

## Responsible Mission Directorate:

Space Technology Mission Directorate (STMD)

## Responsible Program:


Small Business Innovation Research/Small Business Tech Transfer


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## Project Transitions

 **January 2010:** Project Start

 **July 2010:** Closed out

**Closeout Summary:** High Chamber Pressure, Light Weight Thrusters, Phase I Project Image

**Closeout Documentation:**

- Final Summary Chart Image(<https://techport.nasa.gov/file/139046>)

## Project Management

**Program Director:**

Jason L Kessler

**Program Manager:**

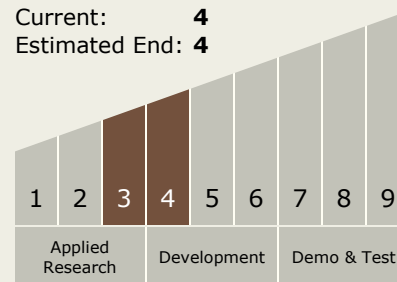
Carlos Torrez

**Principal Investigator:**

Timothy Mckechnie

## Technology Maturity (TRL)

Start: **3**  
Current: **4**  
Estimated End: **4**



## Technology Areas

**Primary:**

- TX01 Propulsion Systems
  - └ TX01.1 Chemical Space Propulsion
    - └ TX01.1.3 Cryogenic

## Target Destinations

Earth, The Moon, Others Inside the Solar System, Outside the Solar System, The Sun, Mars